

N93-26910



Solar System Exploration Division:
Requirements for Space Nuclear Propulsion

Nuclear Propulsion Technical Interchange
Meeting

Sandusky, OH
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**SSSED REQUIREMENTS
FOR SPACE NUCLEAR
PROPULSION**

Topics



- Solar System Exploration Goals and Missions
- Nuclear Electric Propulsion Rationale
- Nuclear Electric Propulsion Requirements
- Low-Power Missions
- Summary

SSED REQUIREMENTS FOR SPACE NUCLEAR PROPULSION	Solar System Exploration Goals	
<ul style="list-style-type: none"> • Solar System Origins <ul style="list-style-type: none"> – Understand the Process of Solar System Formation, in Particular Planetary Formation, and the Physical and Chemical Evolution of Protoplanetary Systems. • Planetary Evolution and State <ul style="list-style-type: none"> – Obtain an In-Depth Understanding of the Planetary Bodies in Our Solar System and Their Evolution Over the Age of the Solar System. • Evidence of Life <ul style="list-style-type: none"> – Search for Evidence of Life in Our Own and Other Planetary Systems, and Understand the Origin and Evolution of Life on Earth and Other Planets. • Robotic and Human Exploration <ul style="list-style-type: none"> – Conduct Scientific Exploration of the Moon and Mars, and Utilize the Moon as a Base of Scientific Study in Participation with NASA's Mission from Planet Earth. 		

SSED REQUIREMENTS FOR SPACE NUCLEAR PROPULSION	Next Mission Phases Outer Planets	
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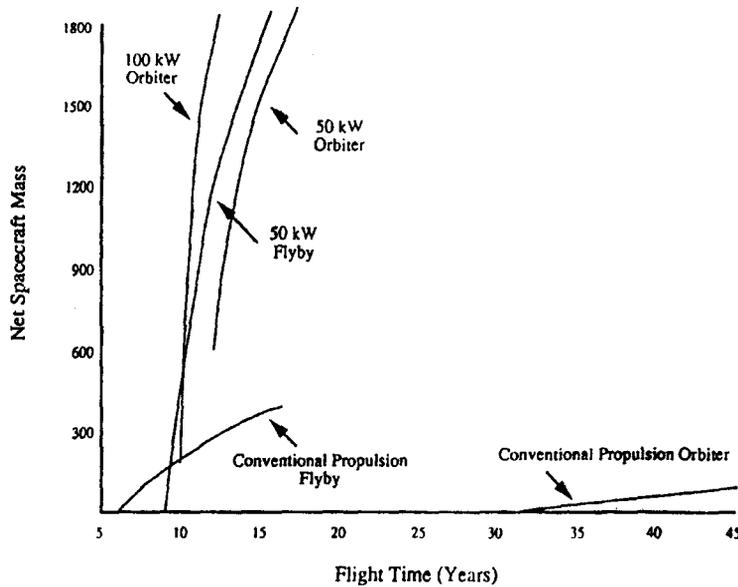
	Outer Planets					Other Planetary Systems
	Jupiter	Saturn	Uranus	Neptune	Pluto	Extrasolar
<i>Reconnaissance</i>	Pioneer 10 Flyby Pioneer 11 Flyby Voyager 1 Flyby Voyager 2 Flyby	Pioneer 11 Flyby Voyager 1 Flyby Voyager 2 Flyby	Voyager 2 Flyby	Voyager 2 Flyby	Flyby	Toward Other Planetary Systems
<i>Exploration</i>	Galileo Orbiter/ Jupiter Probe	Cassini Orbiter/ Titan Probe	Orbiter/ Probe	Orbiter/ Probe		
<i>Intensive Study</i>	Jupiter Grand Tour					



- Nuclear Reactor Heat Source, Ion Propulsion System
 - Much More Efficient than Chemical Propulsion
- NEP Required for Next-Generation Outer Solar System Missions
 - Provides Payload Capability Unobtainable With Conventional Propulsion
 - Reduces Flight Time, Launch Vehicle Requirements
 - Also Enables High-Power Science Experiments
- SP100 Technology Baseline
 - Capable of 100KW for Outer Planet Missions
 - Lifetimes Up to 10 Years (Full Power)
 - Compatible With Active Power Conversion Technologies



PLUTO MISSION PERFORMANCE



SSED REQUIREMENTS FOR SPACE NUCLEAR PROPULSION	Jupiter Grand Tour	
<p><u>Science Objectives</u></p> <ul style="list-style-type: none"> • Thorough Characterization of Galilean Satellites <ul style="list-style-type: none"> - Geology, Morphology, Elemental Composition - Gravitational and Magnetic Properties - Interactions with Jupiter's Magnetosphere • Follow-On to Galileo Study of Jupiter <ul style="list-style-type: none"> - Atmosphere, Inner Magnetosphere, Ring System <p><u>NEP Mission Capabilities</u></p> <ul style="list-style-type: none"> • Sequential Orbiting of All 4 Galilean Satellites <ul style="list-style-type: none"> - Comprehensive Imaging and Spectroscopy - Radar Sounding, Altimetry, Other Active Experiments • Possible Addition of Jupiter Polar Orbiter or Satellite Landers • Large Science Payload, ≈ 10 Year Mission Duration <ul style="list-style-type: none"> - Conventional Propulsion: 4 Separate Launches 		

SSED REQUIREMENTS FOR SPACE NUCLEAR PROPULSION	Multiple Main Belt Asteroid Rendezvous	
<p><u>Science Objectives</u></p> <ul style="list-style-type: none"> • Comprehensive Study of Asteroid Physical Characteristics <ul style="list-style-type: none"> - Size, Shape, Density, Spin Properties - Surface Composition, Solar Wind Interactions • Variations With Solar Distance • Meaningful Sample Size, Variety of Spectral Types <p><u>NEP Mission Capabilities</u></p> <ul style="list-style-type: none"> • Rendezvous With 4-6 Main Belt Asteroids <ul style="list-style-type: none"> - Approximately 60 Days at Each Target - Possible Intervening Slow Flybys - Unlimited Orbit-Change Capability • Large Science Payload <ul style="list-style-type: none"> - Imaging, Spectroscopy, Radiometry - Multiple Penetrators • Total Mission Duration ≈ 10 Years <ul style="list-style-type: none"> - Conventional Propulsion: Max. 2 Targets, > 8 Years Duration 		

SSED REQUIREMENTS FOR SPACE NUCLEAR PROPULSION	NEP System Requirements for Planetary Missions			
Mission	Power Level (kWe)	Lifetime Full/Mission (Yrs)	Specific Mass (kg/kWe)	Technology Need Dates (year)
Far Outer Planet Orbiters/Probes	100	7-9/14-15	< 35	- 2000-01
Jupiter Grand Tour	100	8/11	< 35	- 2000
Multiple Mainbelt Asteroid Rendezvous	90	7/10	< 35	- 2000
Comet Nucleus Sample Return	90	4/8	< 35	- 2003

SSED REQUIREMENTS FOR SPACE NUCLEAR PROPULSION	Low-Power NEP Missions			
<ul style="list-style-type: none"> • Initial NEP System Will Address Reduced Requirements <ul style="list-style-type: none"> - Simplifies Development, Reduces Cost - Still Capable of Excellent Planetary Missions • Mission/System Studies Ongoing <ul style="list-style-type: none"> - Joint NASA/DOE Report Issued - JPL/LeRC Study Focussing on Low-Power Missions • Preliminary Mission Options Include: <ul style="list-style-type: none"> - Mars Orbiter, Phobos-Deimos Rendezvous (SEI Focus) - Main-Belt Asteroid Missions - Jupiter Satellite Mission - Solar Probe • System Requirements (Preliminary): <ul style="list-style-type: none"> - Minimum 20 kWe NEP System - Minimum 3 Year Lifetime (Full Power) - Growth Potential to 100 kWe, 10 Years Lifetime 				



- Nuclear Electric Propulsion Enables Next-Generation Outer Solar System Mission
- Requirements
 - 100 kWe, 10 -yr. Lifetime (Full-Power), < 35 kg/kWe
 - Initial System: > 20 kWe, > 3 Yr. Full-Power Lifetime
 - Full-Power System Launch ~2005